

A Computational Fluid Dynamics Study of Fluid Catalytic Cracking Cyclones

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Fluidized Catalytic Cracking (FCC)

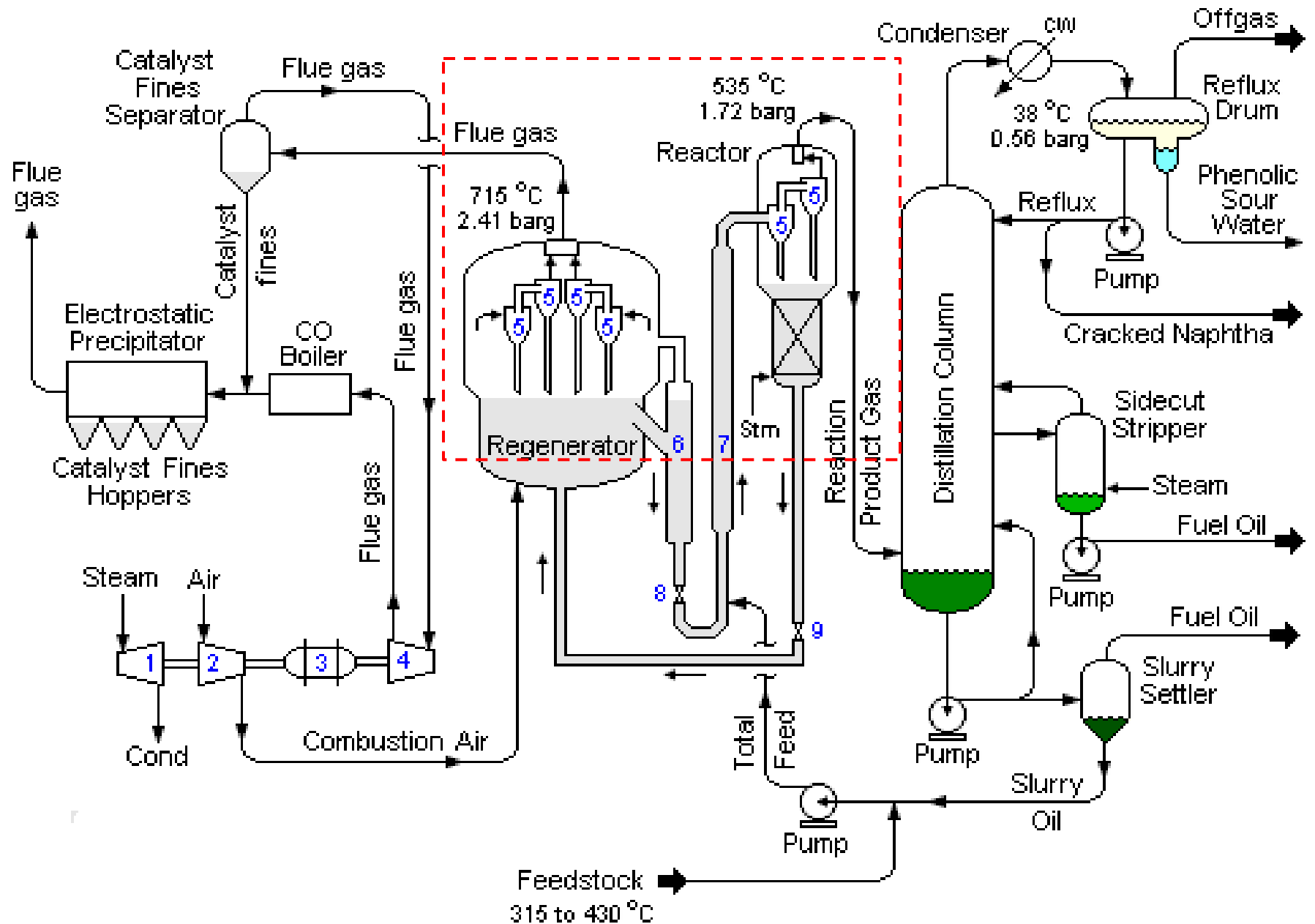
Reactor

- Vaporized oil feedstock is exposed to a catalyst heated to approximately 1,000 deg F - causing the cracking reaction.
- Usually takes place in a riser pipe
- **Cyclones are used to separate the reacted oil feedstock from the catalyst.**
- Reaction is endothermic.

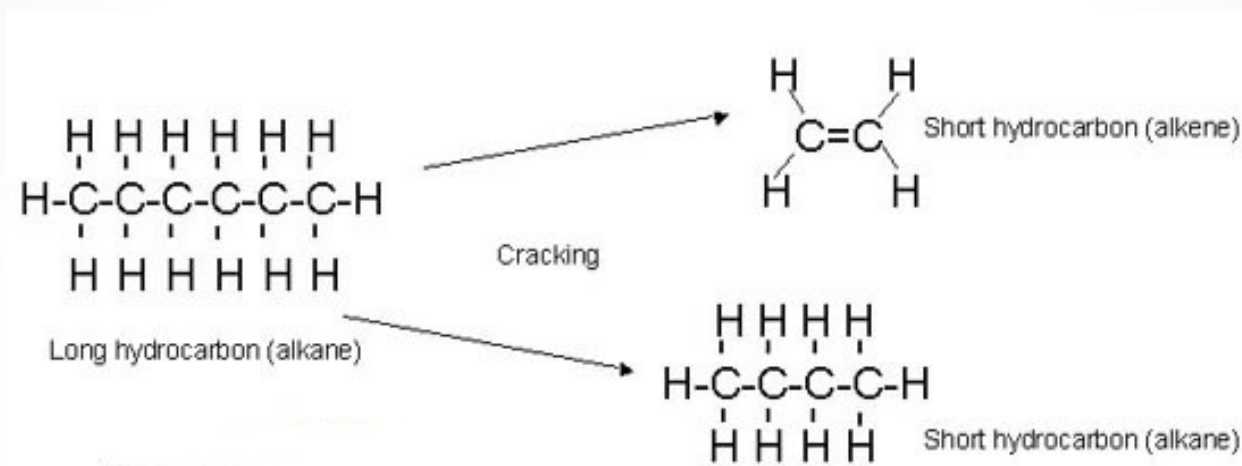
Regenerator

- Coke is built up on the catalyst during the reaction.
- The catalyst is exposed to oxygen in the regenerator igniting combustion.
- The catalyst is "regenerated" by the combustion of the coke.
- **Cyclones are used to separate the catalyst from the flue gas.**
- Reaction is exothermic.

FCC Refinery Flow Diagram



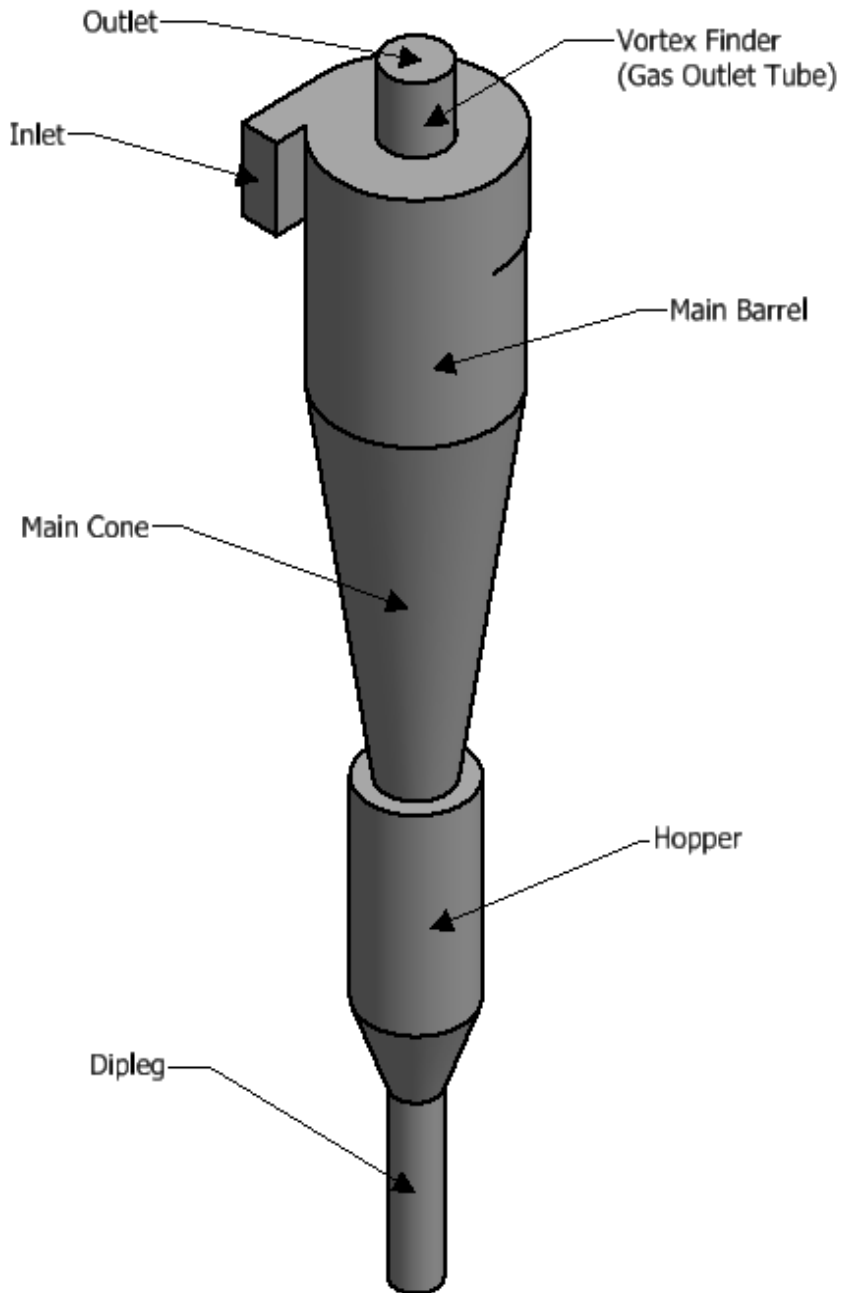
Chemistry of Cracking



- Alkanes (paraffins) are long chain single bonded hydrocarbons.
- Cracking breaks the long chain into usable short chain alkanes and alkenes.
 - alkanes used for gasoline and fuel oil
 - alkenes have a double covalent bond: ethanol, alcohol, and plastics

Reactor and Regenerator

- Considered the heart of a refinery.
- Reactor and Regenerator are thermally balanced.
 - heat generated from the exothermic reaction in the regenerator, supplies the endothermic reaction in the reactor.
- Catalyst used to facilitate the reaction is expensive and abrasive.
- Shutdowns are required to repair cyclone lining from erosion.

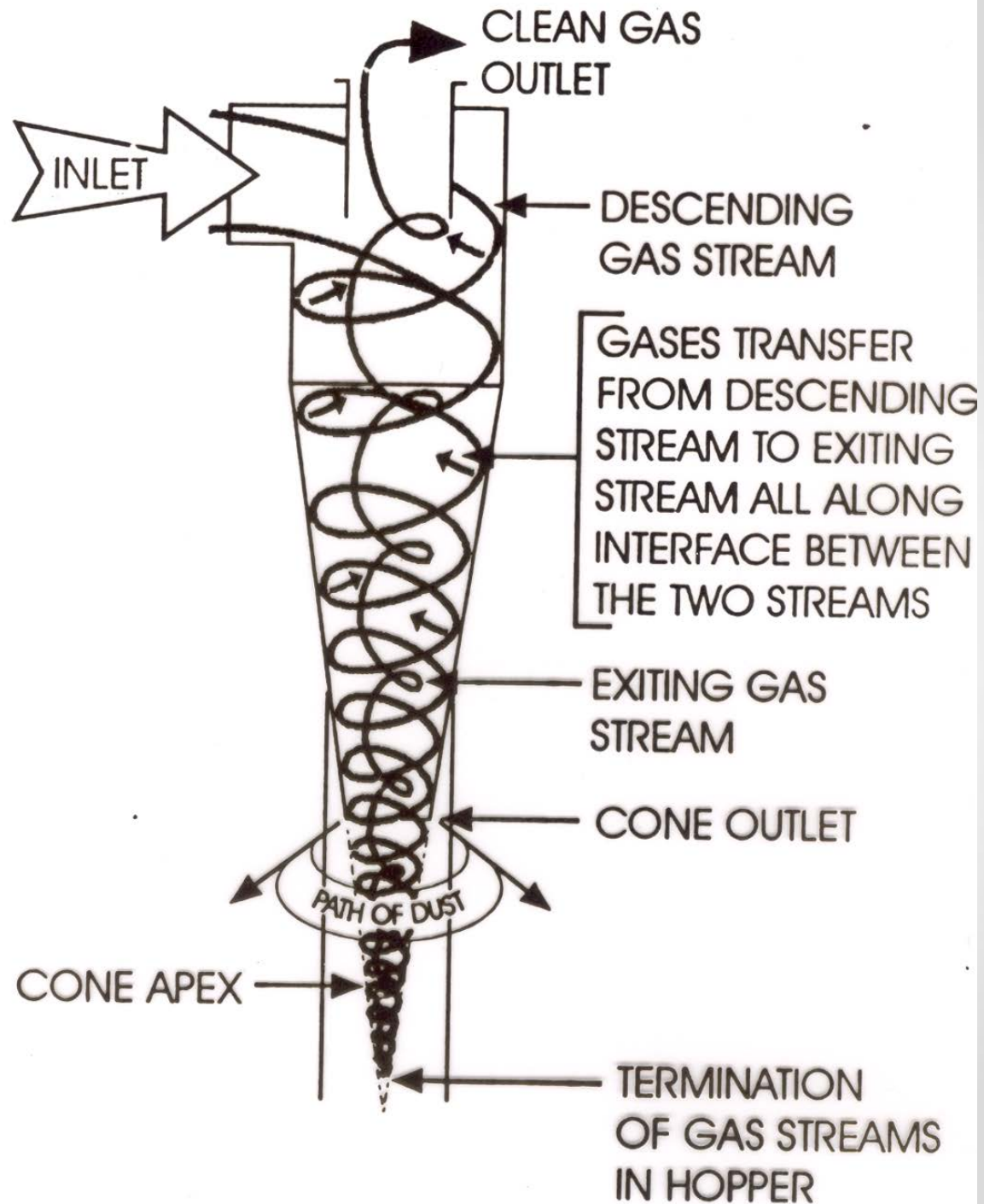


Cyclone Nomenclature

FCC Cyclone Installation



Cyclone Basics



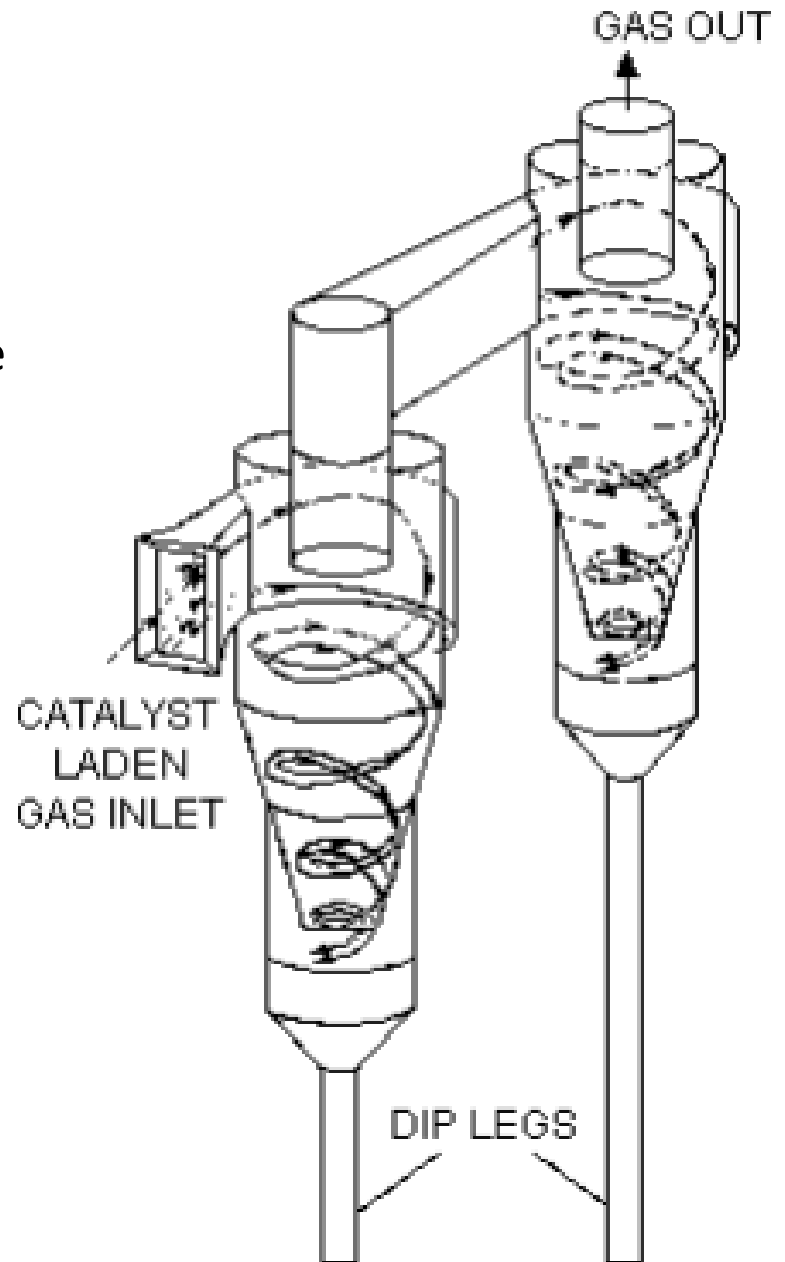
2 Stage Cyclone System

Dense Flow Cyclones (High Mass Loading)

- Large quantities of particulate enter the cyclone.
- 1st Stage Cyclone.

Dilute Flow Cyclones (Low Mass Loading)

- Loading is approximately 5% of a dense flow cyclone.
- 2nd Stage Cyclone



Three Major Parameters of Cyclone Performance

1. Efficiency

- Percent of particles captured of the particles that enter.
- Typical ranges from 95% to 99%.

2. Pressure Drop

- Pressure loss due to cyclone operation.
- Cost to operate system.

3. Erosion

- Catalyst is highly abrasive.
- Maintenance costs.
- Downtime.

Methods of Cyclone Design

- 1) **Laboratory Modeling**
- 2) **Algebraic Equations**
- 3) **Computational Fluid Dynamics (CFD)**

Laboratory Modeling

- Laboratory modeling and empirical testing have been the major methods for developing cyclone technology.
- Scale models were created and results verified by industrial performance.
 - Particulate tests
 - Pressure Loss
 - Erosion

Algebraic Models

- Developed from empirical data.
- Rigid equations that can only be used for narrow applications.

Model	Equation	Remarks
Shepherd and Lapple [42]	$\xi_g = \frac{16ab}{D_c^2} \quad (35)$	Tangential inlet; ambient air conditions
Alexander [11]	$\xi_g = 4.62 \left(\frac{ab}{D_c D_e} \right) \left[\left(\left(\frac{D_c}{D_e} \right)^{2n} - 1 \right) \left(\frac{1-n}{n} \right) + f_g \left(\frac{D_c}{D_e} \right)^{2n} \right] \quad (36)$	Experiments with scroll and tangential inlets
	$f_g = 0.8 \left[\frac{1}{n(1-n)} \left(\frac{4-2^{2n}}{3} \right) - \left(\frac{1-n}{n} \right) \right] + 0.2 \left[(2^{2n} - 1) \left(\frac{1-n}{n} \right) + 1.5(2^{2n}) \right] \quad (37)$	Air and combustion gases, up to 1100 °C
	$n = 1 - (0.67 D_c^{0.14}) \left(\frac{T}{283} \right)^{0.8} \quad (4)$	
Barth [13]	$\xi(\lambda = \lambda_g) = \left(\frac{ab}{\pi D_c^2 / 4} \right)^2 (\xi_b + \xi_e) \quad (38)$	
	Loss in the cyclone body $\xi_b = \frac{D_e}{D_c} \left(\frac{1}{(v_{ze}/v_{te} - ((H-S)/(0.5D_e))\lambda)^2} - \left(\frac{v_{te}}{v_{ze}} \right)^2 \right) \quad (39)$	
	Loss in the vortex finder $\xi_e = K \left(\frac{v_{te}}{v_{ze}} \right)^{4/3} + \left(\frac{v_{te}}{v_{ze}} \right)^2 \quad (40)$	3.41 < K < 4.4
Muschelknautz and Kambrock [43]	$\xi(\lambda = \lambda_g) = \left(\frac{ab}{\pi D_c^2 / 4} (\xi_b + \xi_e) \right) \quad (41)$	Tangential and scroll inlets
	$\xi_b = \lambda \frac{A_S}{0.9V} \frac{\rho_g}{2} (v_{tw} v_{te})^{1.5} \quad (42)$	Flow field based on Barth's model [13]
	$\xi_b = 2 + 3 \left(\frac{v_{te}}{v_{ze}} \right)^{4/3} + \left(\frac{v_{te}}{v_{ze}} \right)^2 \quad (43)$	Ambient P, T conditions $\lambda = \lambda_g \approx 0.006$ A_S is the total inner area of cyclone contributing to friction
Casal et al. [44]	$\xi_g = 11.3 \left(\frac{ab}{D_c^2} \right)^2 + 2.33 \quad (44)$	Comparative study of six correlations

Computation Fluid Dynamics (CFD) Models

- Flow is represented by the Navier-Stokes equation.
 - Coupled differential equations that must be solved simultaneously.
- The Navier-Stokes equation is not closed for turbulent flow.
 - Approximations, averages, and constants from testing are used to solve the equations for turbulent flow.

Turbulent Flow Models

1. **k- ϵ**

- k – turbulent kinetic energy
- ϵ - dissipation rate of k
- Assumes isotropic turbulence
- Fast and consistent convergence

2. **Renormalized Group (RNG) k- ϵ**

- Adds an additional variable to the standard k- ϵ model

Turbulent Flow Models

3. Reynolds **S**tress **M**odel (RSM)

- Incorporates transport equations for each Reynolds stress dissipation component
- Forgoes isotropic turbulence assumption
- 7 equations
- High computation costs

4. Large **E**ddy **S**imulation (LES)

- Transient

Cyclones



Cyclone 1

standard design



Cyclone 2

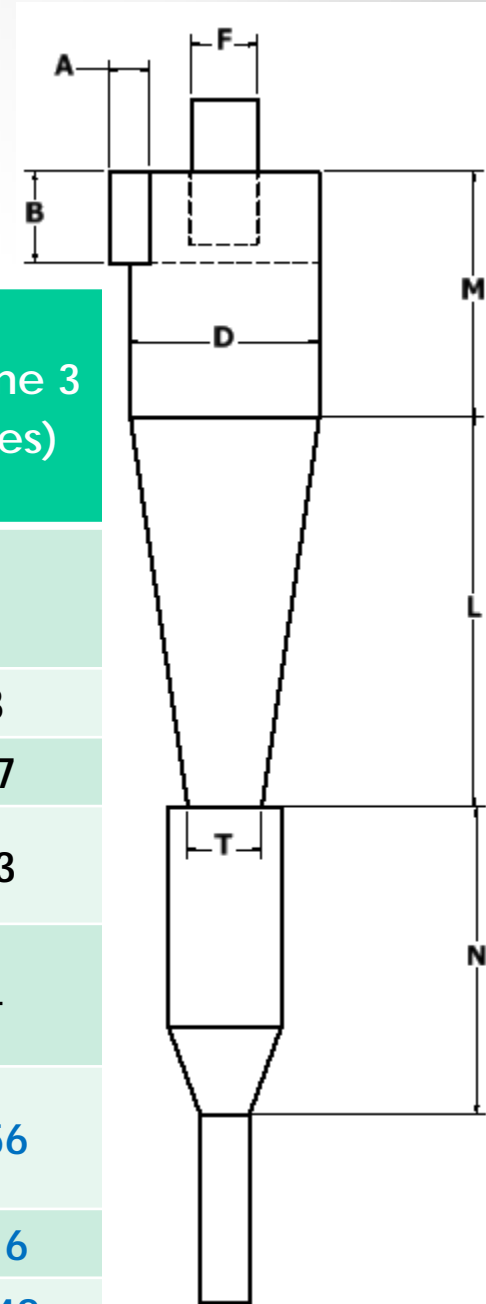
modified outlet tube



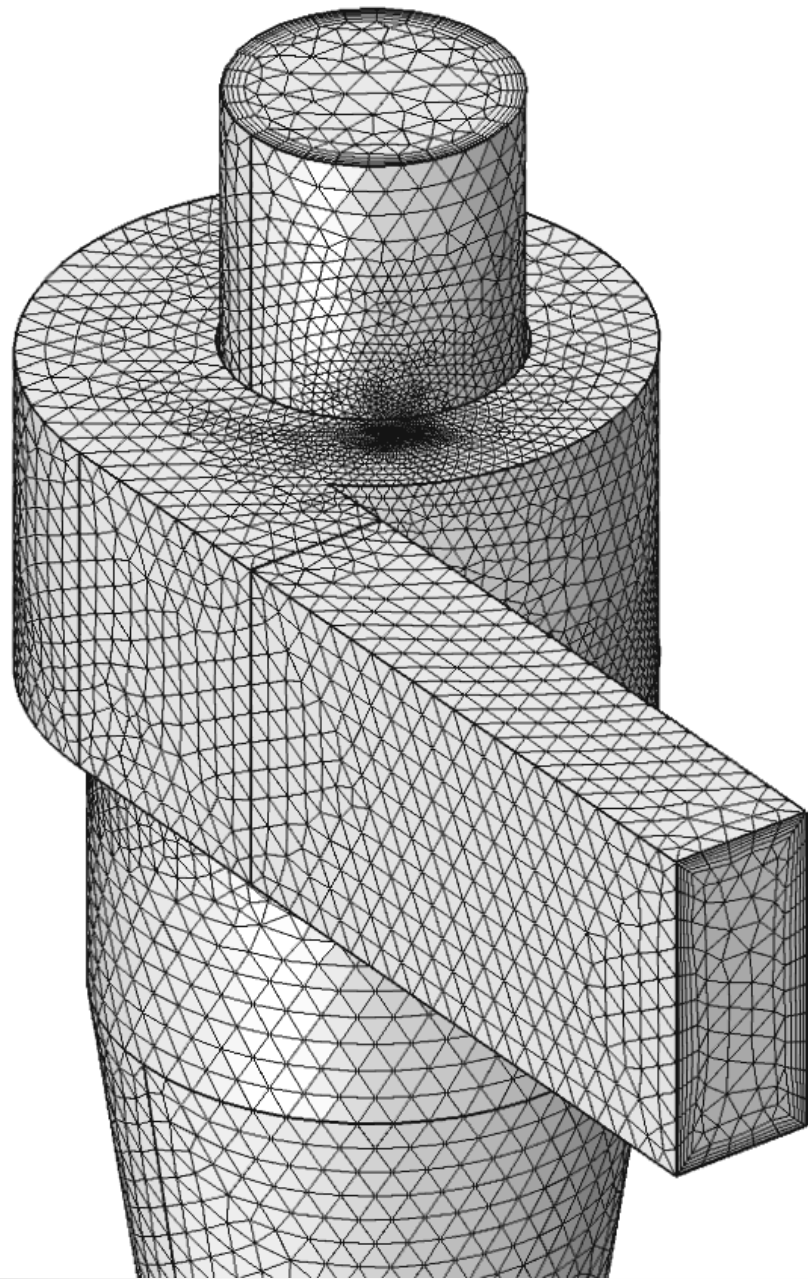
Cyclone 3

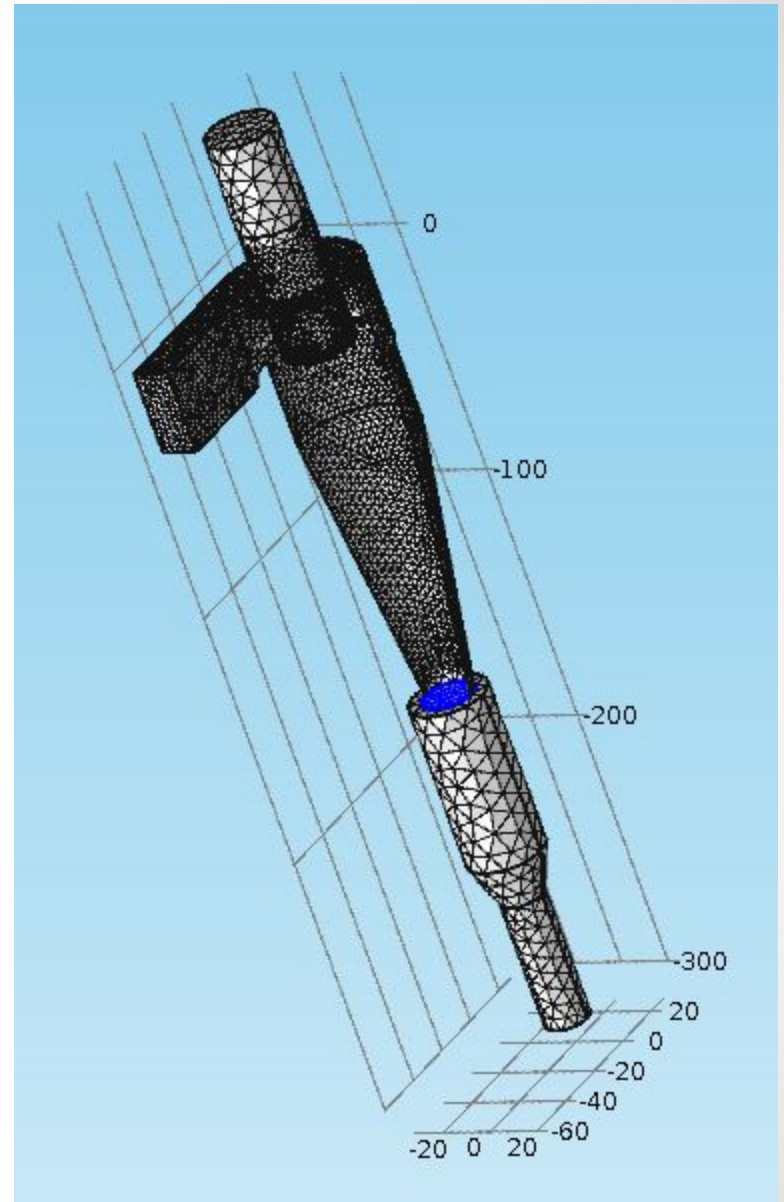
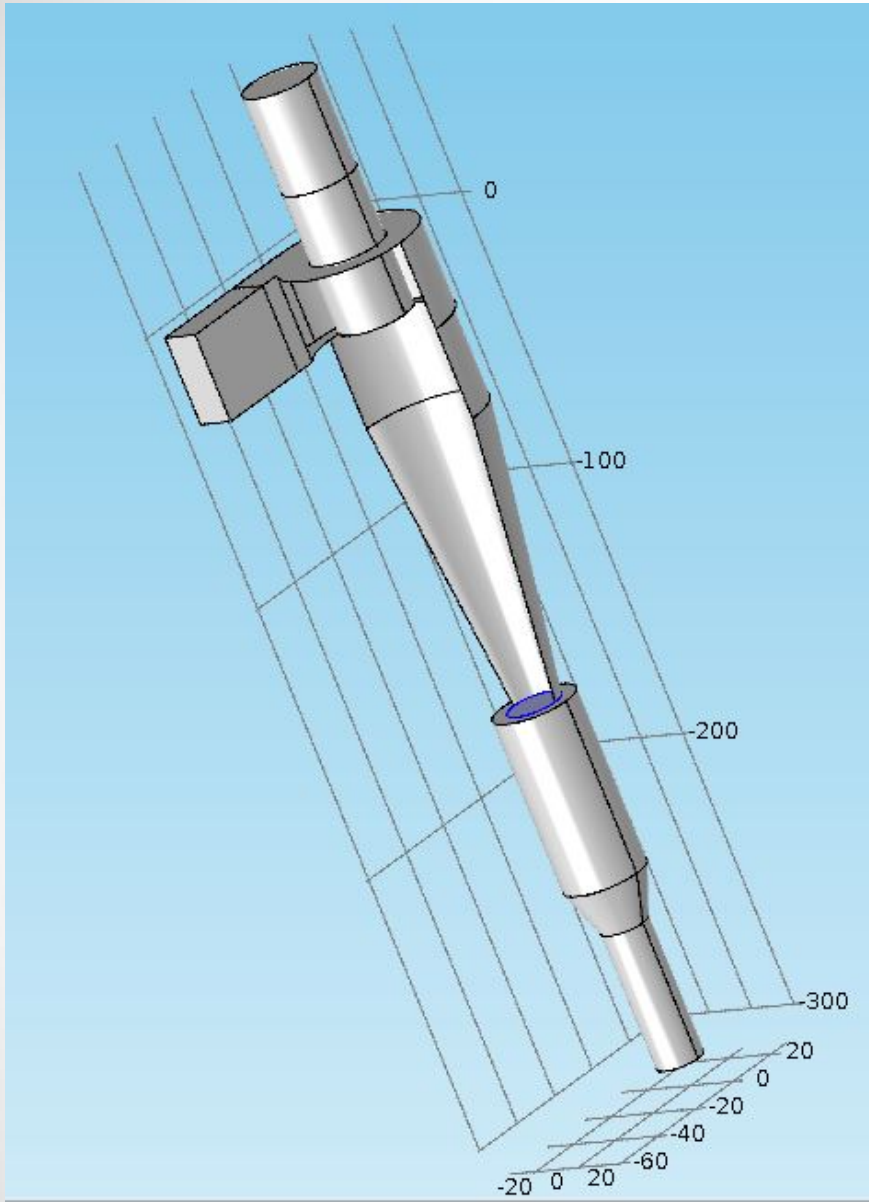
high efficiency cyclone

Cyclone Sizing and Dimensions



Description	Variable	Model Input	Cyclone 1 (inches)	Cyclone 2 (inches)	Cyclone 3 (inches)
Barrel Diameter	D	D	50	50	66
Barrel Height	M	M/D	1.3	1.3	1.3
Cone Height	L	L/D	2.07	2.07	2.07
Hopper Height	N	N/D	1.63	1.63	1.63
Cone Diameter	T	T/D	0.4	0.4	0.4
Vortex Finder Diameter	F	F/D	0.535	0.35	0.356
Inlet Width	A	A/D	0.285	0.285	0.216
Inlet Height	B	B/D	0.64	0.64	0.4848



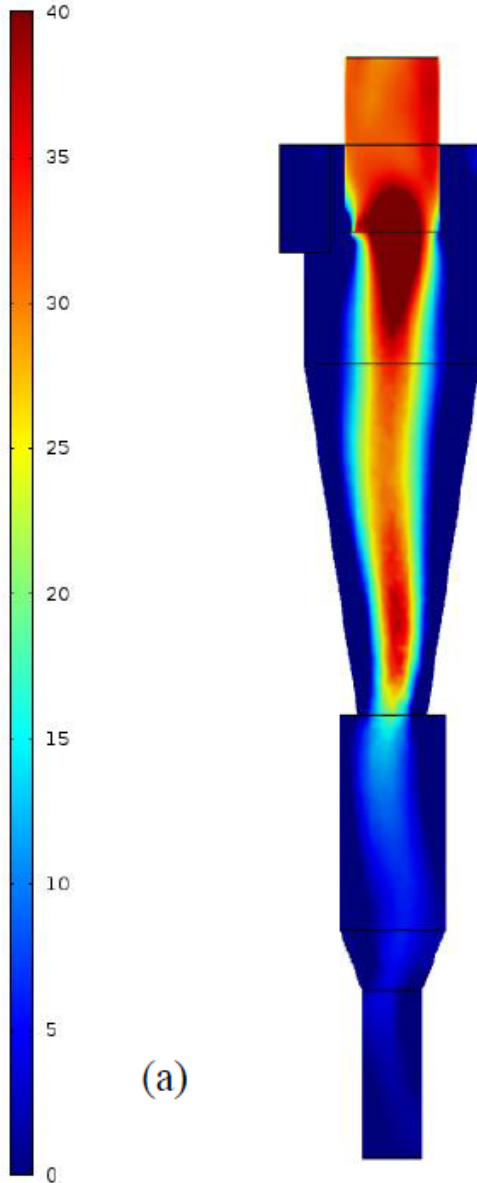


Vortex Structure

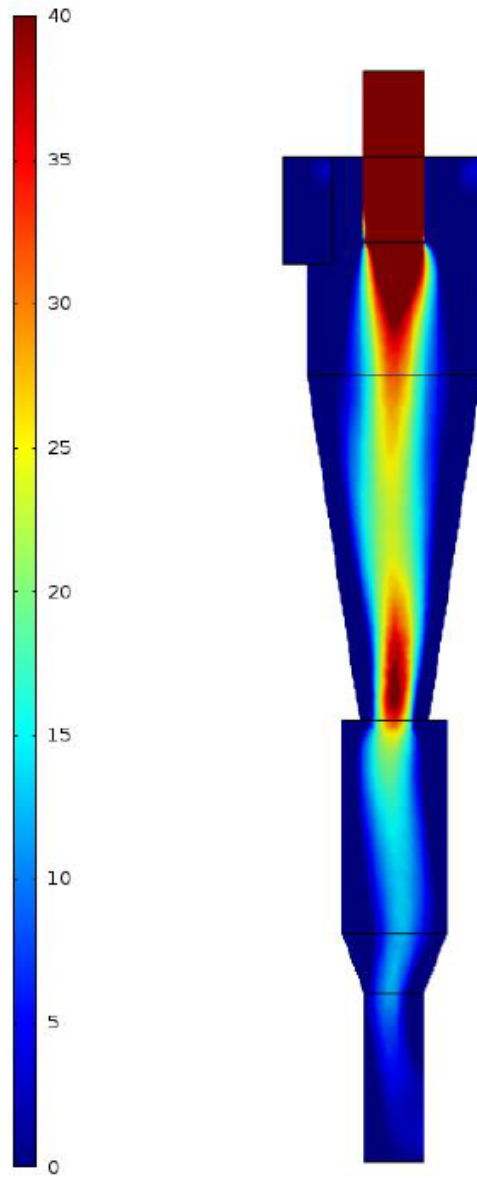
- Can provide useful information regarding erosion and collection efficiency.
- There is not an exact apex or termination point of the vortex.
 - Subjective to how the profile is portrayed
 - Vortex can dissipate after entry into the hopper
- Following the gas path can give clues about the particle path.
- Vortex length is related to erosion in the hopper and dipleg.



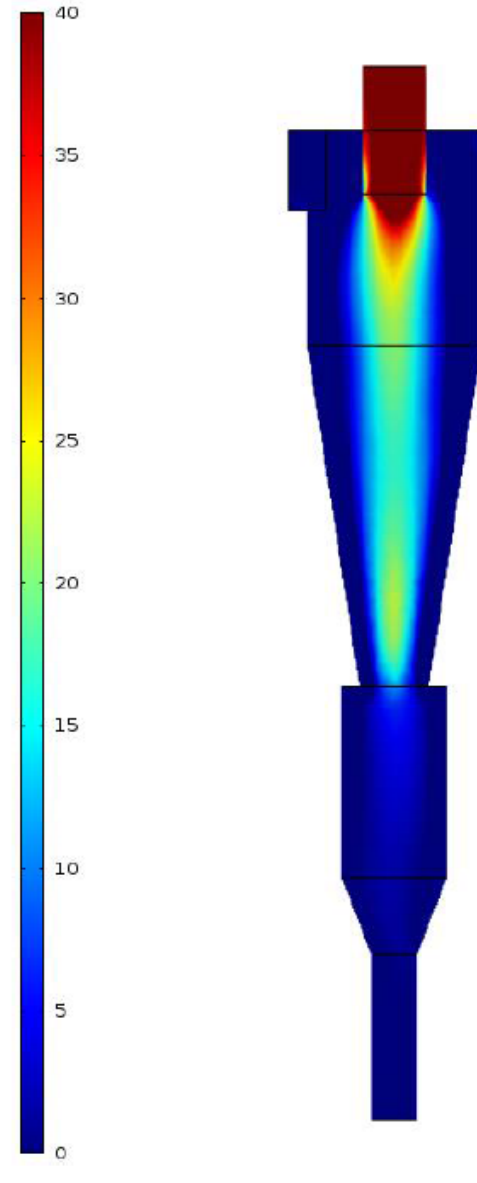
Cyclone 1



Cyclone 2



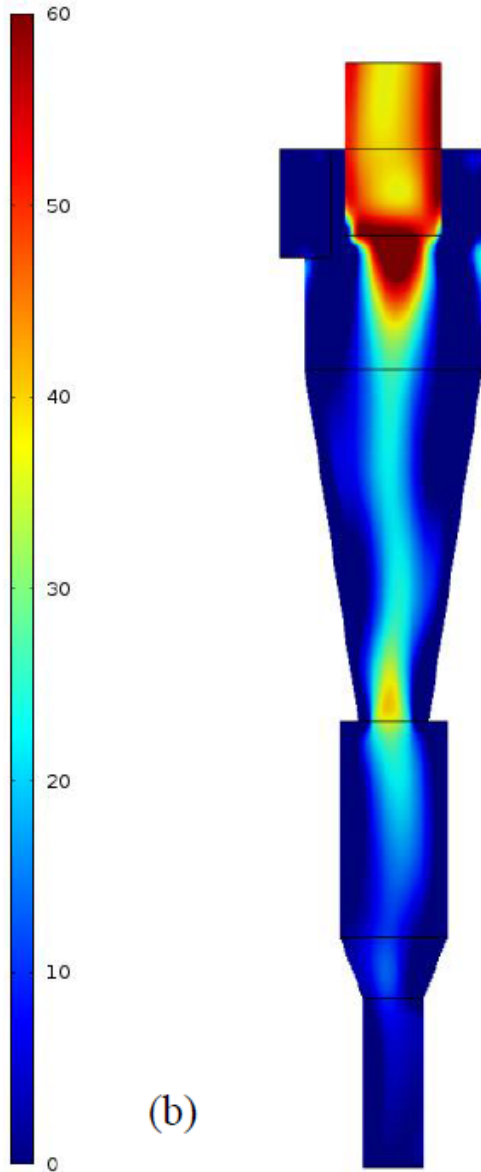
Cyclone 3



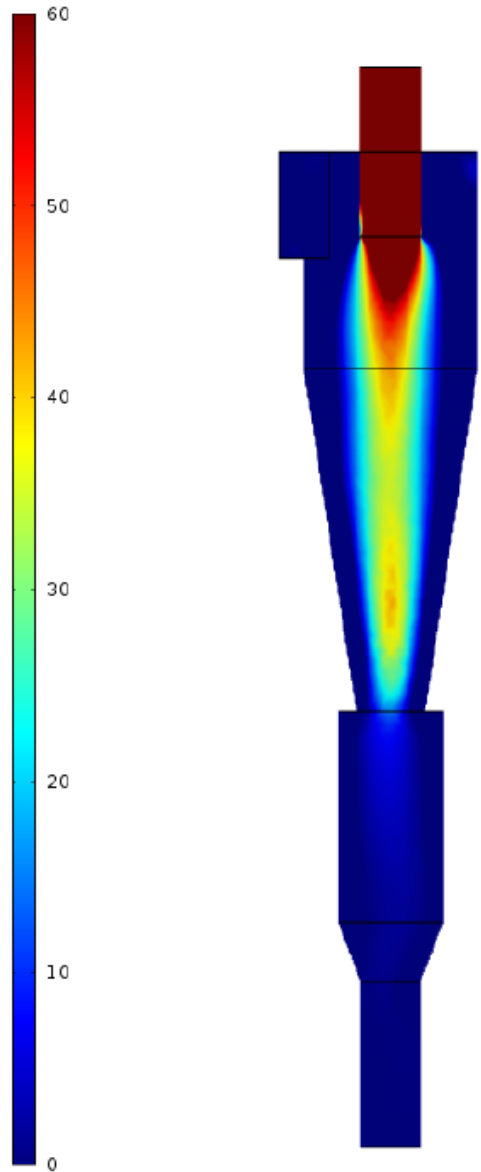
(a)

Axial velocity at (a) 40 fps inlet

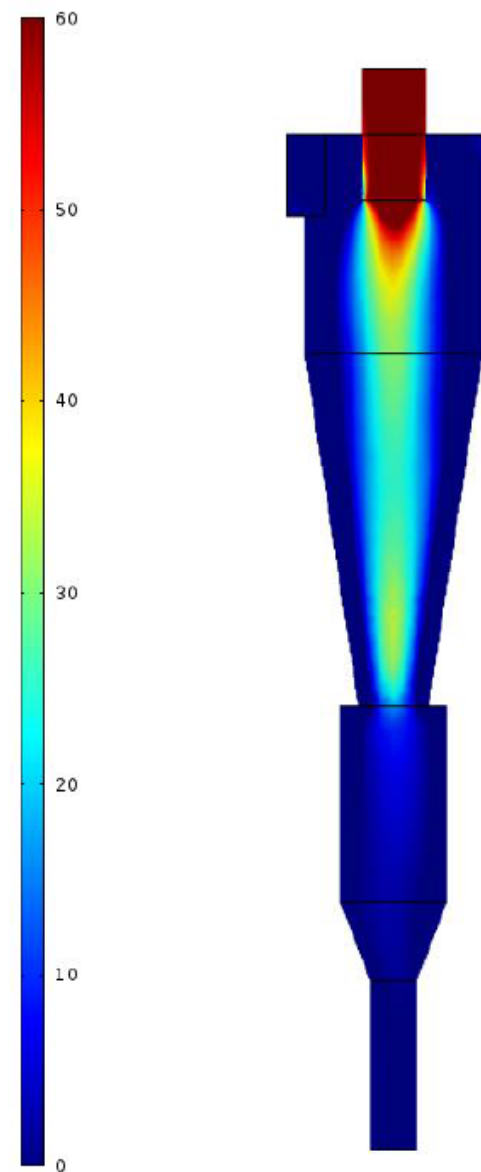
Cyclone 1



Cyclone 2



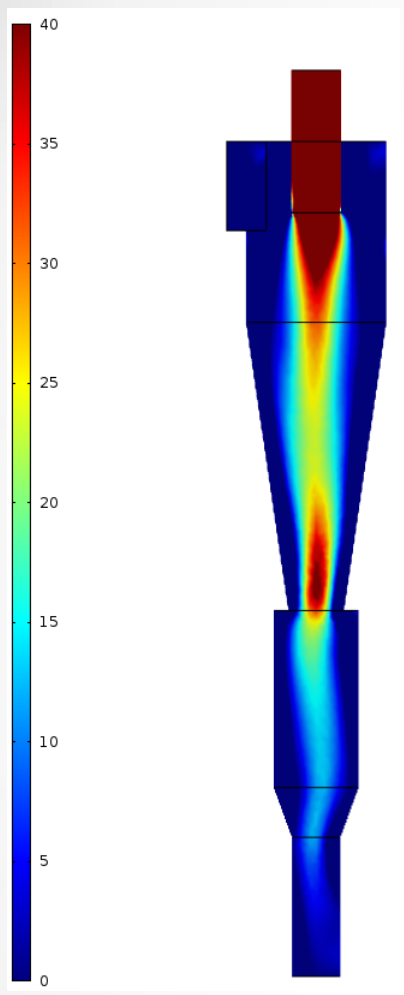
Cyclone 3



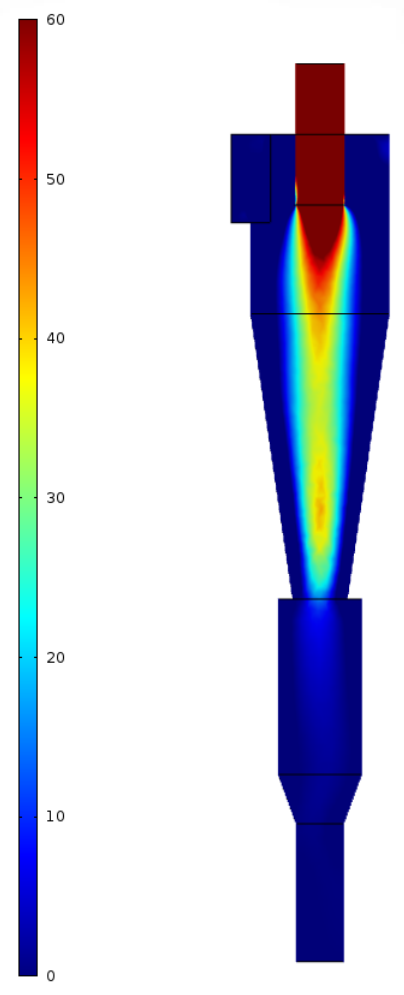
(b)

Axial velocity at (b) 60 fps inlet.

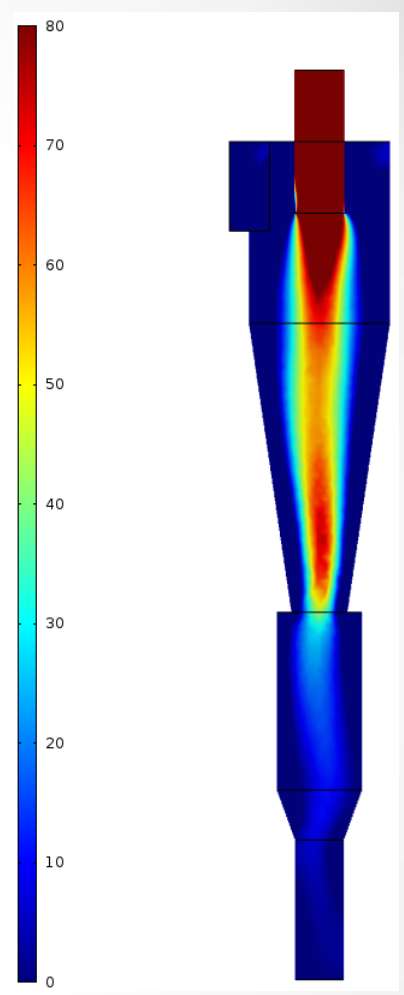
Cyclone 2: Axial Velocity



40fps



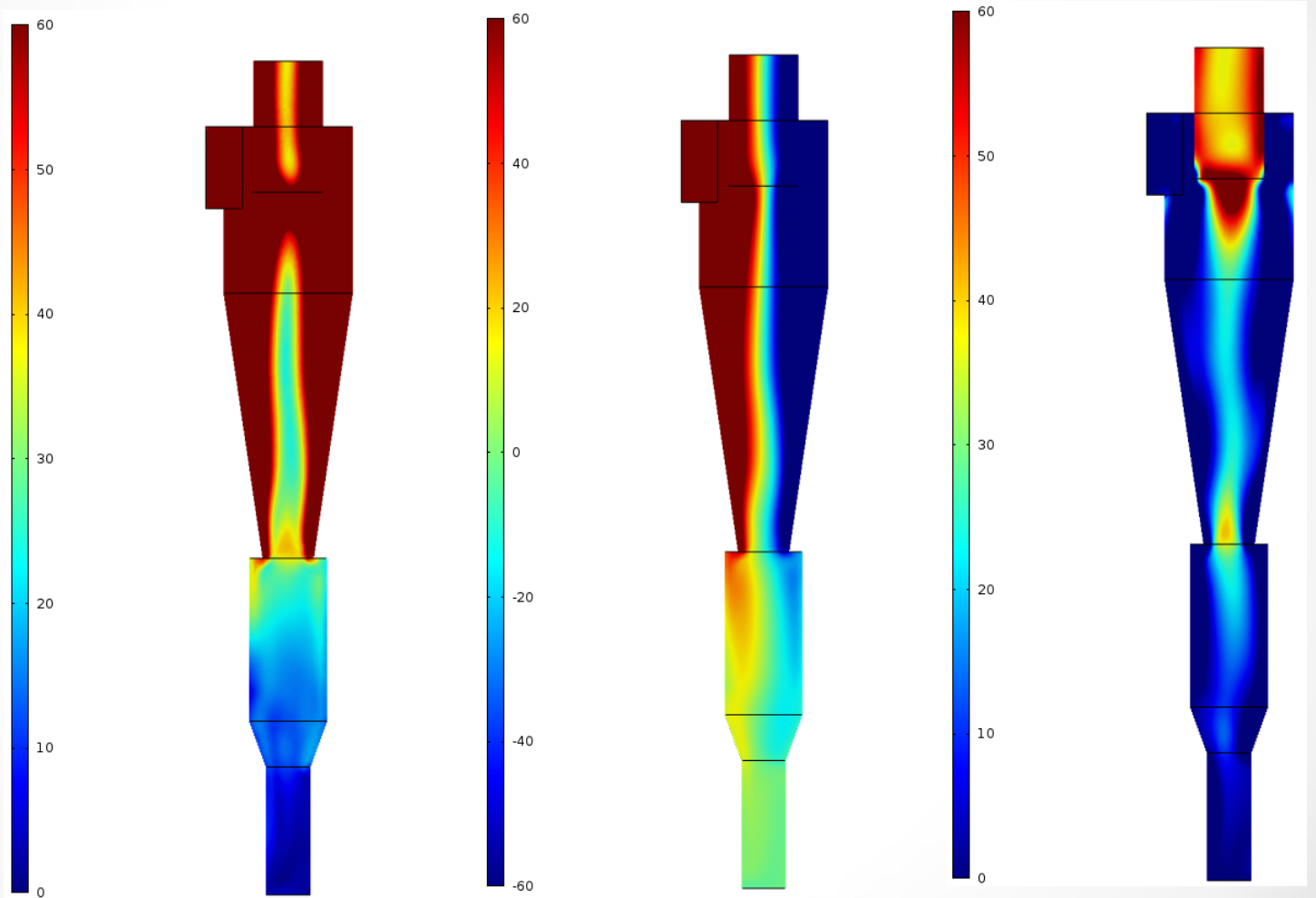
60fps



80fps

Velocity: Cyclone 1

60fps Inlet



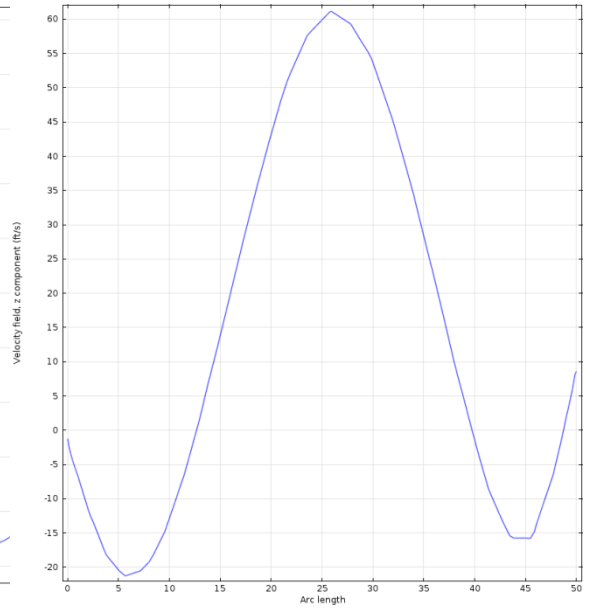
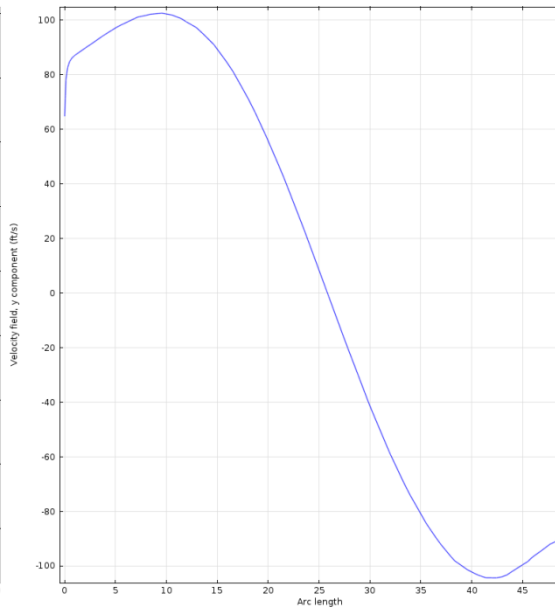
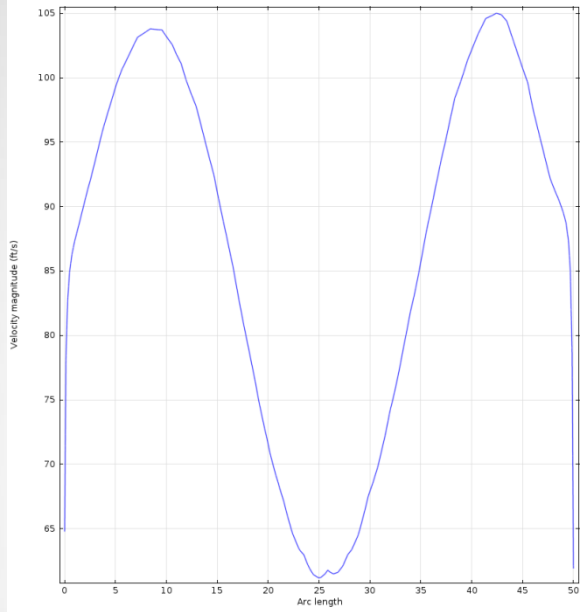
Velocity Magnitude

Tangential Velocity

Axial Velocity

Cyclone 1

at 40 inches below cyclone roof, 60fps inlet velocity

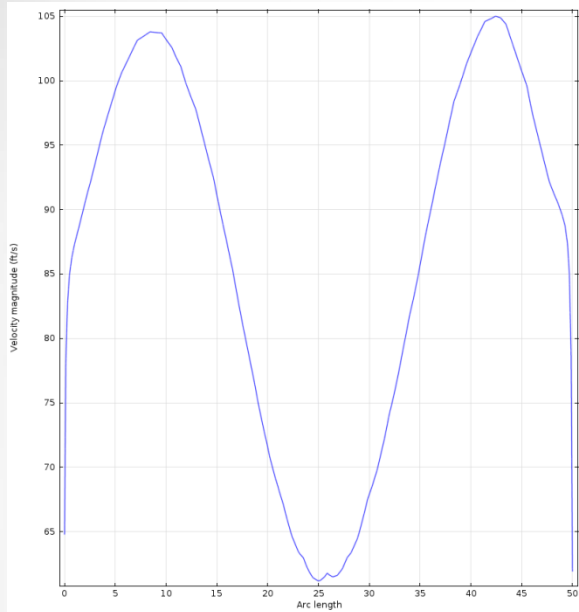


Velocity magnitude

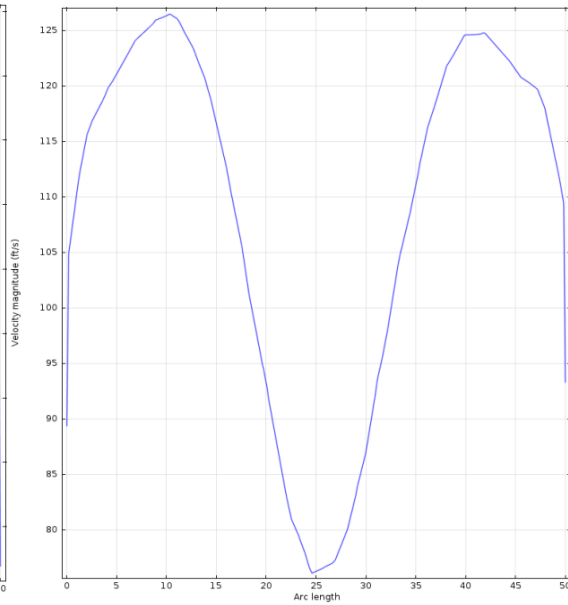
– Tangential velocity

– Axial velocity

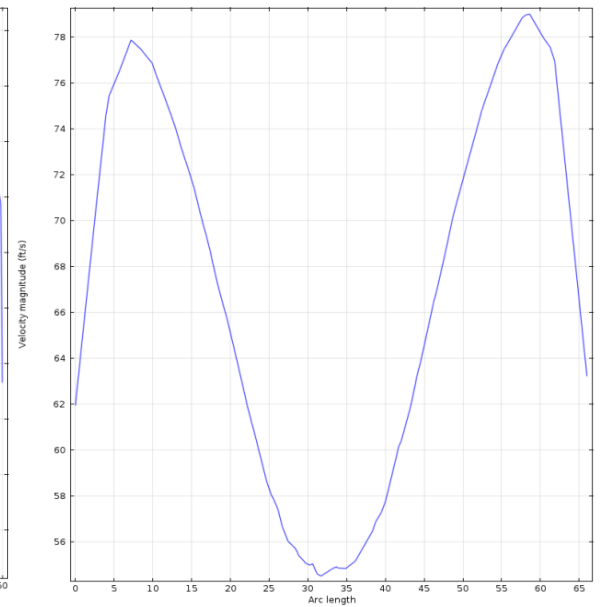
Cyclone 1



Cyclone 2



Cyclone 2

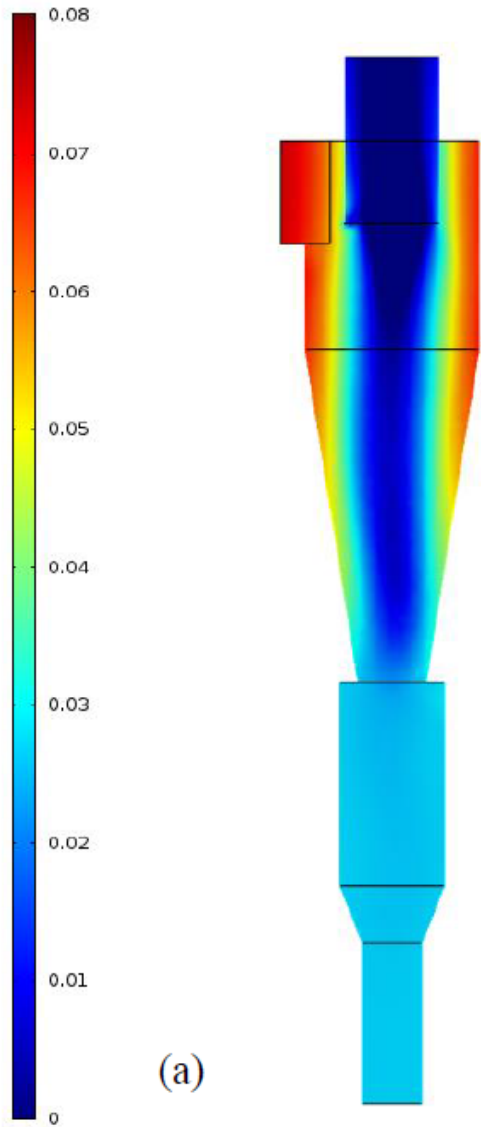


Velocity magnitude at 40 inches below cyclone roof, 60fps inlet velocity

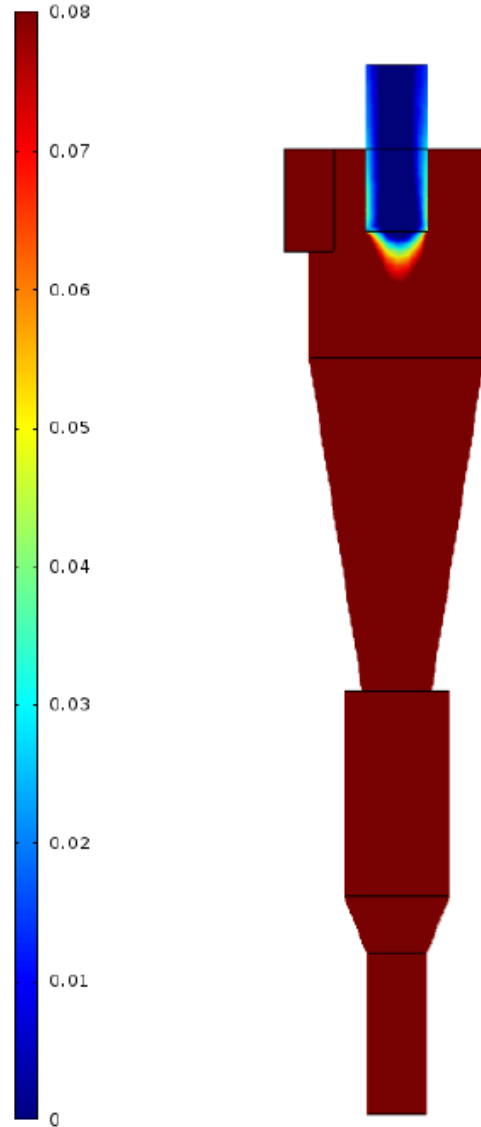
Pressure Loss

- Pressure loss of a cyclone is often inversely related to cyclone efficiency.
- Increasing the inlet velocity of a cyclone will often increase the efficiency, but at the cost of increasing the pressure loss.
- It has been determined that approximately 80% of the pressure loss is due to the viscous stresses.
- The remaining loss of pressure occurs at the inlet and outlet of the cyclone and by frictional forces along the cyclone wall.

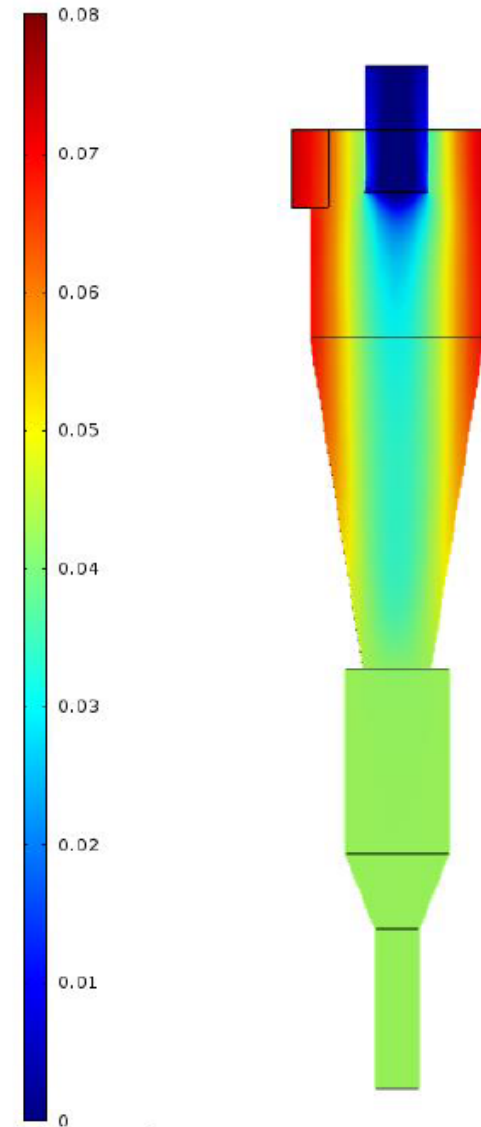
Cyclone 1



Cyclone 2



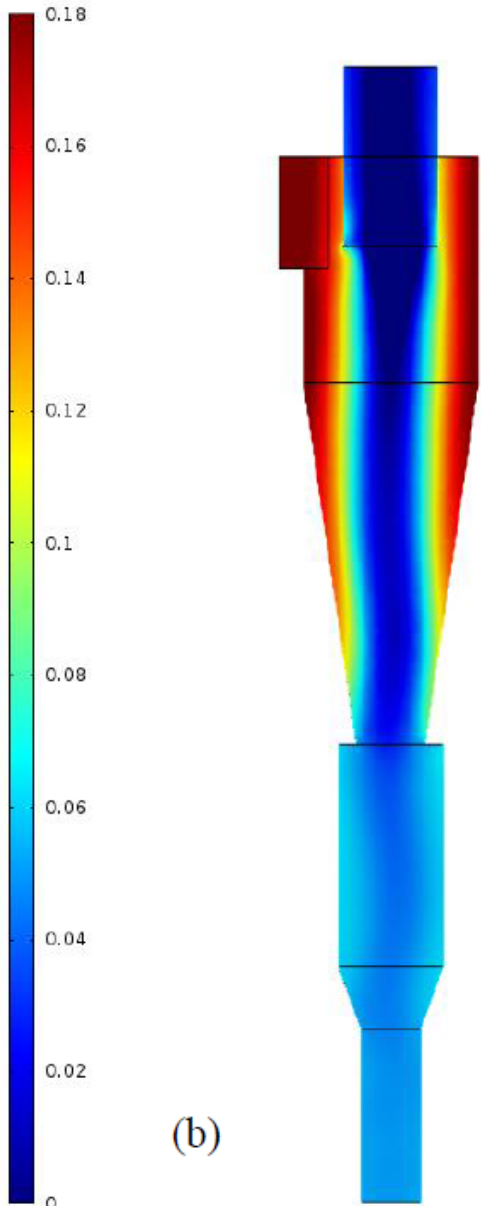
Cyclone 3



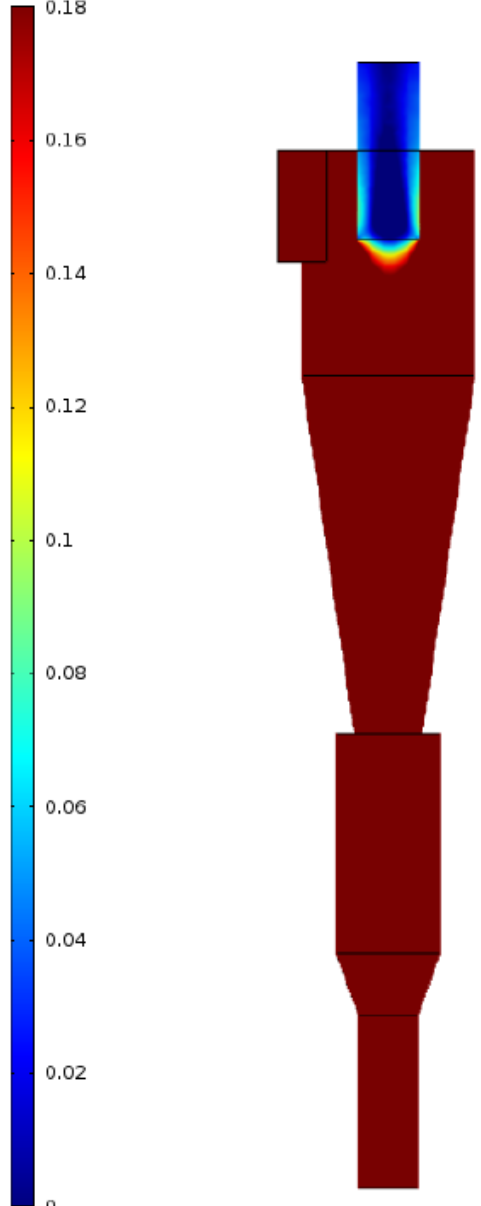
(a)

Static pressure at (a) 40 fps

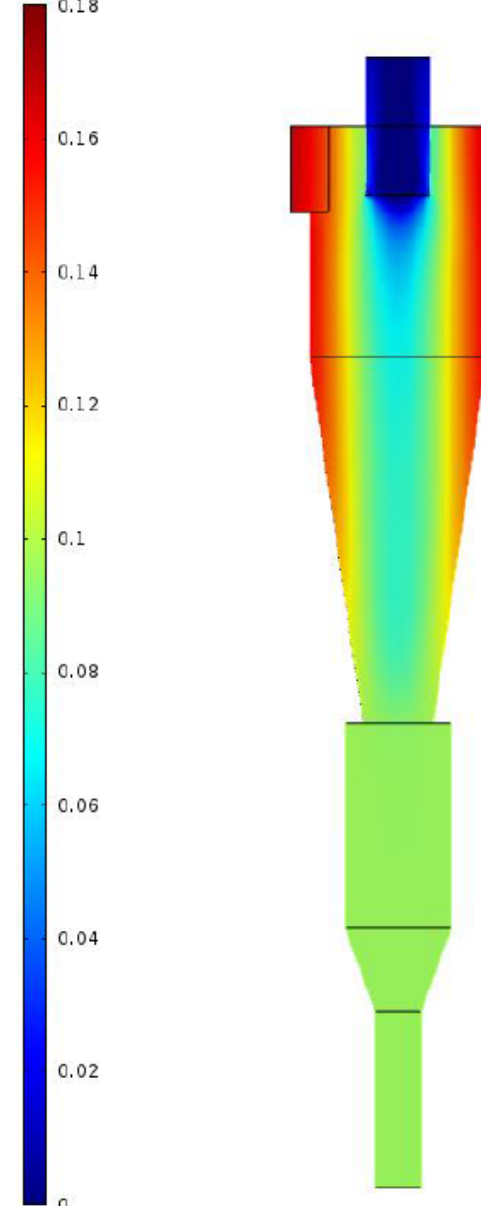
Cyclone 1



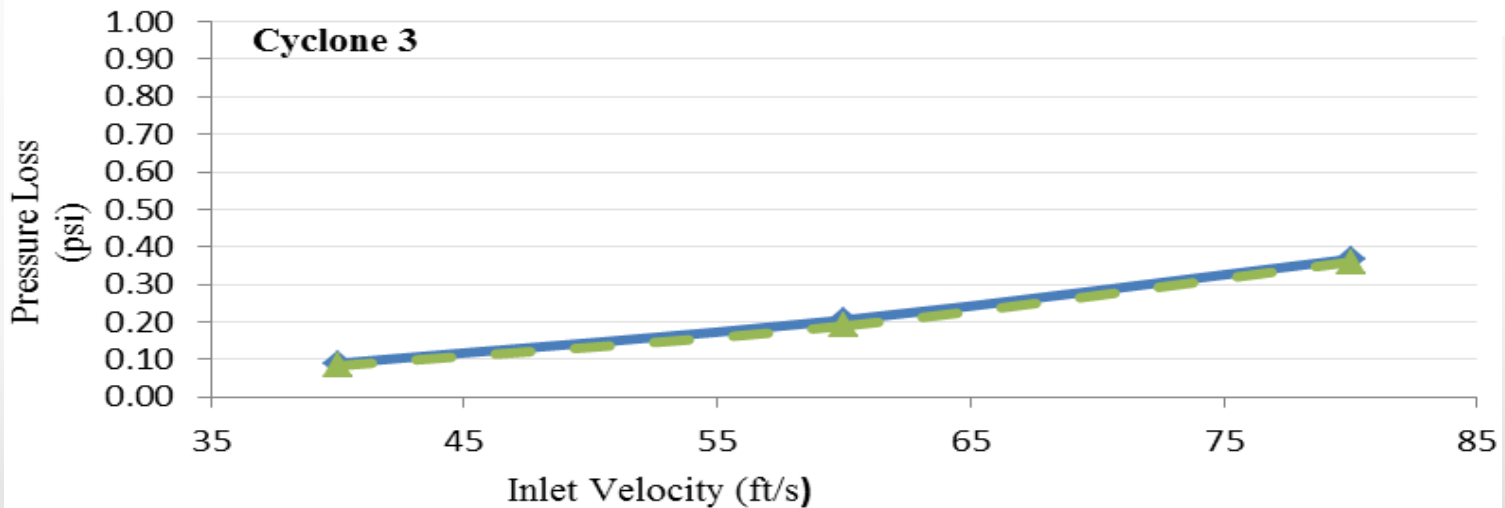
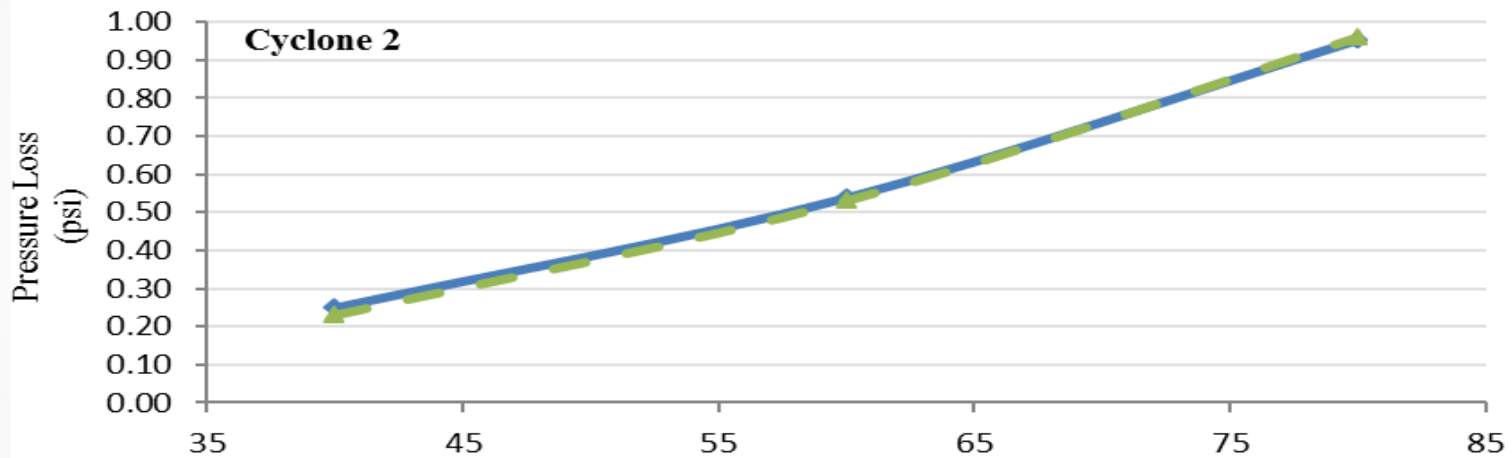
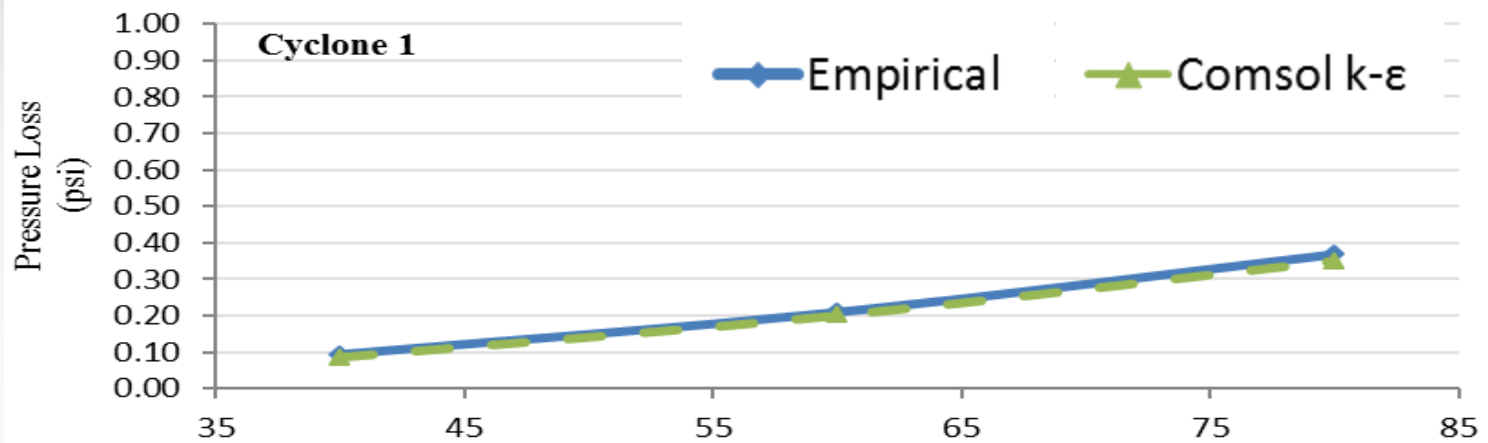
Cyclone 2



Cyclone 3



Static pressure at (b) 60 fps



Pressure Loss

Cyclones 1 and 3 show high pressure regions along the walls. This implies that pressure loss due to drag is dominant.

Cyclone 2 has a modified outlet tube – meaning the outlet area is much less than the inlet area. This increases the outlet velocity.

COMSOL produces pressure loss with an average percent error of about 3% with the error decreasing with increasing inlet velocity.

Conclusions

- The simple geometry of cyclones is deceptive.
 - Turbulent flow is highly complex.
 - Small changes in geometry can produce major effects on performance.
- Cyclones are highly efficient making efficiency increases difficult to detect.
- Model is lacking the effects of anisotropic turbulence

Conclusions

- Cyclones push the limits of CFD code and solvers.
 - Anisotropic turbulence requires very robust and computationally costly solvers.
 - Need to model particle influence on the gas stream
- All parameters of cyclone performance must be considered during design.